

Computation the bow shock at $M=4$ with a 5th-order WENO scheme

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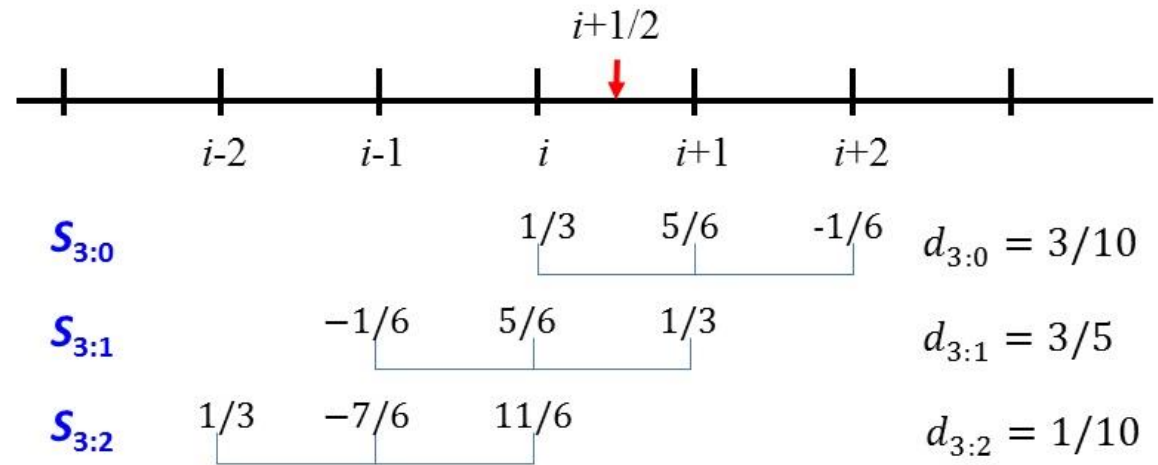
Embry-Riddle Aeronautical University
Daytona Beach FL

Spatial discretization: WENO-5

- Weighted average of interpolated values from all candidate stencils

$$\tilde{v}_{i+1/2}^r = \sum_{j=0}^{k-1} c_{rj} v_j$$

$$v_{i+1/2} = \sum_{r=0}^{k-1} \omega_k \tilde{v}_{i+1/2}^r$$



Shu, 2000, NASA/CR-97-206253
 Henric et. al. 2005, *JCP*
 Bogers et. al. 2008, *JCP*

WENO-5

Implicit time integration

➤ Semi-discretized form

$$\frac{\partial Q}{\partial t} = R(Q)$$

➤ Implicit time integration

- Significantly relaxed stability constraints
- Large time steps (limited by physical considerations), reduce the required CPU time
- Main difficulty solution of the resulting nonlinear system of equations

Implicit time integration - 1

- Implicit 1st order backward Euler

$$\frac{Q^{n+1} - Q^n}{\Delta t} = R(Q^{n+1})$$

Nonlinear implicit equations

$$F(Q^{n+1}) = \frac{1}{\Delta t} Q^{n+1} - R(Q^{n+1}) - \frac{Q^n}{\Delta t} = 0$$

Implicit time integration - 2

- S-stage, implicit Runge-Kutta schemes

$$Y_i = Q^n + \Delta t \sum_{j=1}^i a_{ij} R(t_n + c_j \Delta t, Y_j) \quad i = 1, 2, \dots, s$$

$$Q^{n+1} = Q^n + \Delta t \sum_{i=1}^s b_i R(t_n + c_i \Delta t, Y_i)$$

i-stage

$$F(Y_i) = \frac{1}{\Delta t} Y_i - a_{ii} R(t_n + c_i \Delta t, Y_i) - \left[\frac{Q^n}{\Delta t} + \sum_{j=1}^{i-1} a_{ij} R(t_n + c_j \Delta t, Y_j) \right]$$

= 0

Solution of implicit equations

➤ General equations

$$F(Y_i) = \alpha Y_i + \beta R(Y_i) - b = 0$$

➤ Newton iteration

$$Y^{k+1} = Y^k + \delta Y^k$$

$$\frac{\partial F}{\partial Y} \delta Y^k = -F(Y_i^k)$$

➤ Convergence test

$$\|F(Y_i^{k+1})\|_2 < \varepsilon_{atol}$$

$$\|F(Y_i^{k+1})\|_2 < \varepsilon_{rtol} \|F(Y_i^k)\|_2$$

$$\|\delta Y^{k+1}\|_2 < \varepsilon_{stol}$$

Solution of implicit equations

- LU-SGS – linearization w.r.t. Q^n

Jameson and Yoon, 1987, *AIAA J*
Yoon and Jameson, 1988, *AIAA J*

$$\left(\alpha + \beta \frac{\partial R}{\partial q} \right) \delta q^n = b$$

$$LD^{-1}U\delta q^n = b$$

$$L = \alpha I + \beta(D_x^- A^+ - A^- + \dots)$$

$$D = \alpha I + \beta(A^+ - A^- + \dots)$$

$$U = \alpha I + \beta(D_x^+ A^- + A^+ \dots)$$

Dispersion and dissipation errors

- A-stable is NOT sufficient
 - Dispersion and dissipation errors
 - Oscillations of intermediate solutions
 - Numerical instability, increased CPU time

Model equation $\frac{dQ}{dt} = \mu + i\lambda$

↓ Dissipation terms

Amplification
Numerical $G = \frac{Q^{n+1}}{Q^n}$

Amplification
Exact $G_{exact} = e^{(\mu+i\lambda)\Delta t}$

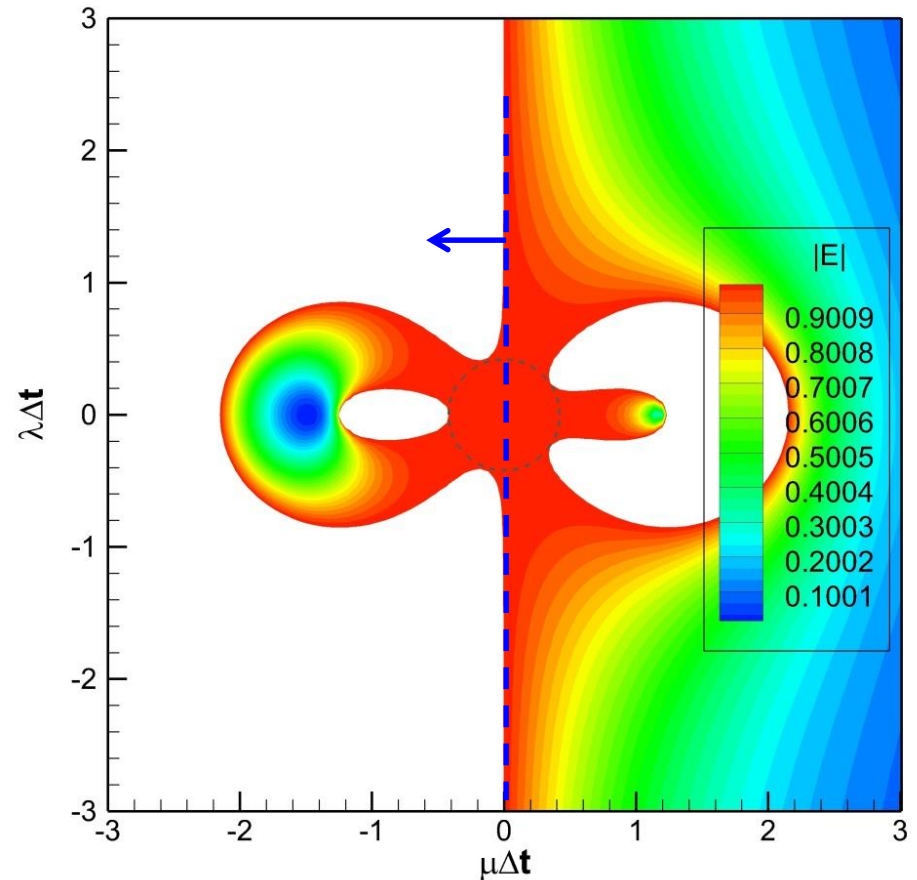
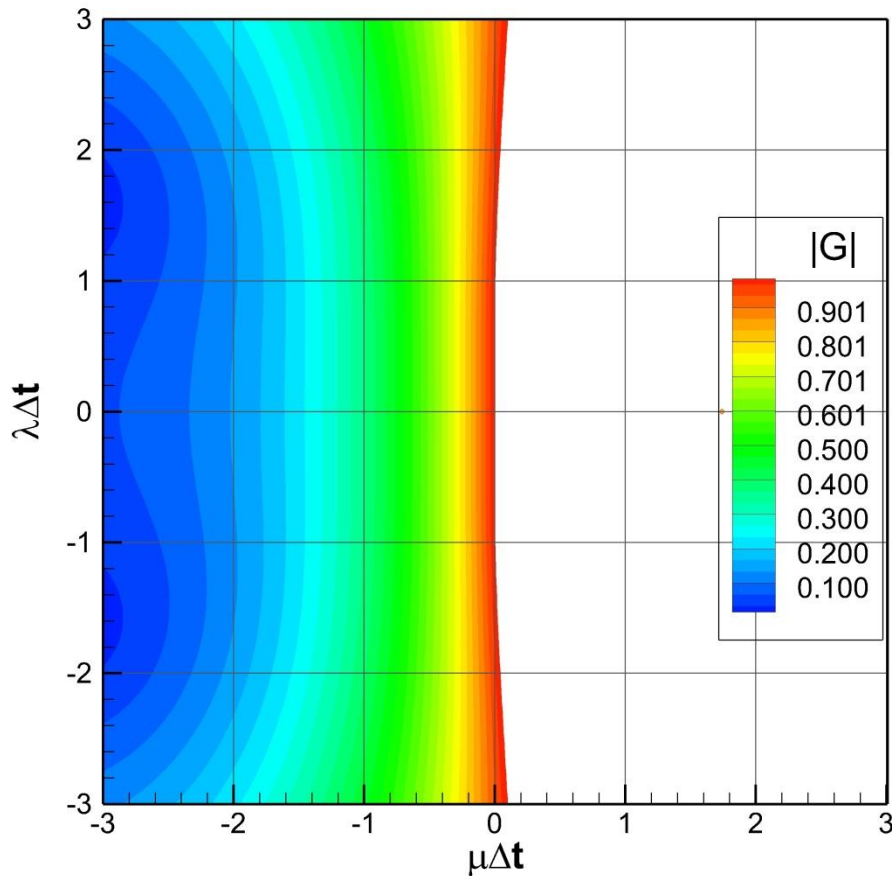
Error $E = \frac{G}{G_{exact}}$

Dispersion and dissipation errors

Narazi et al., JCP, 2015

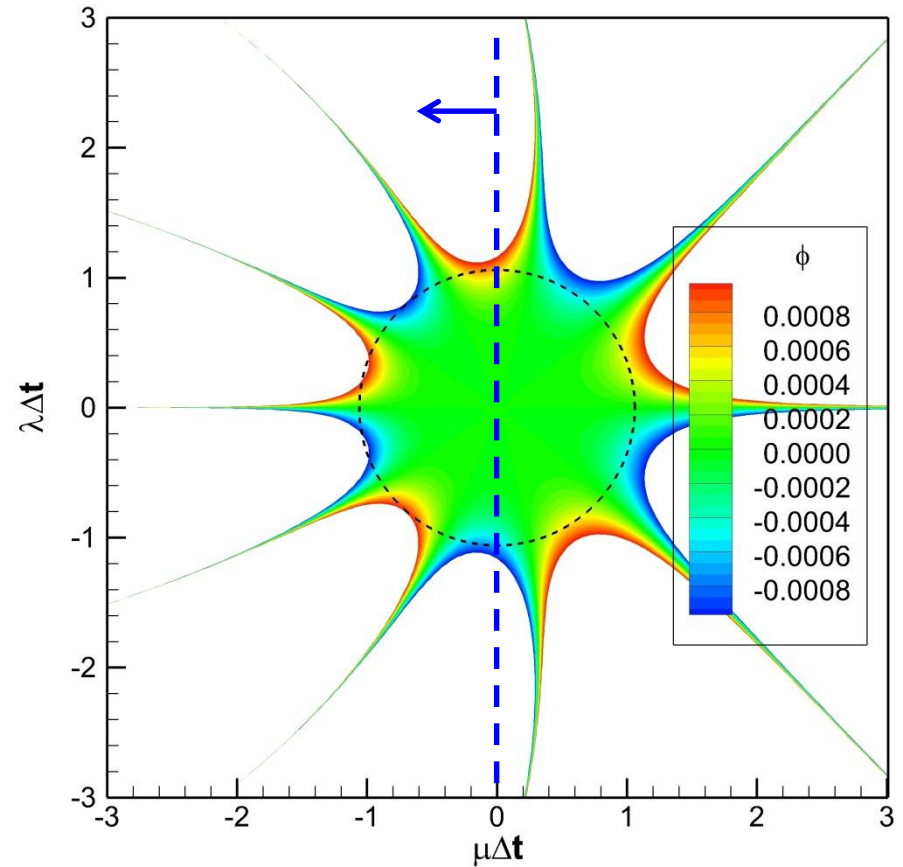
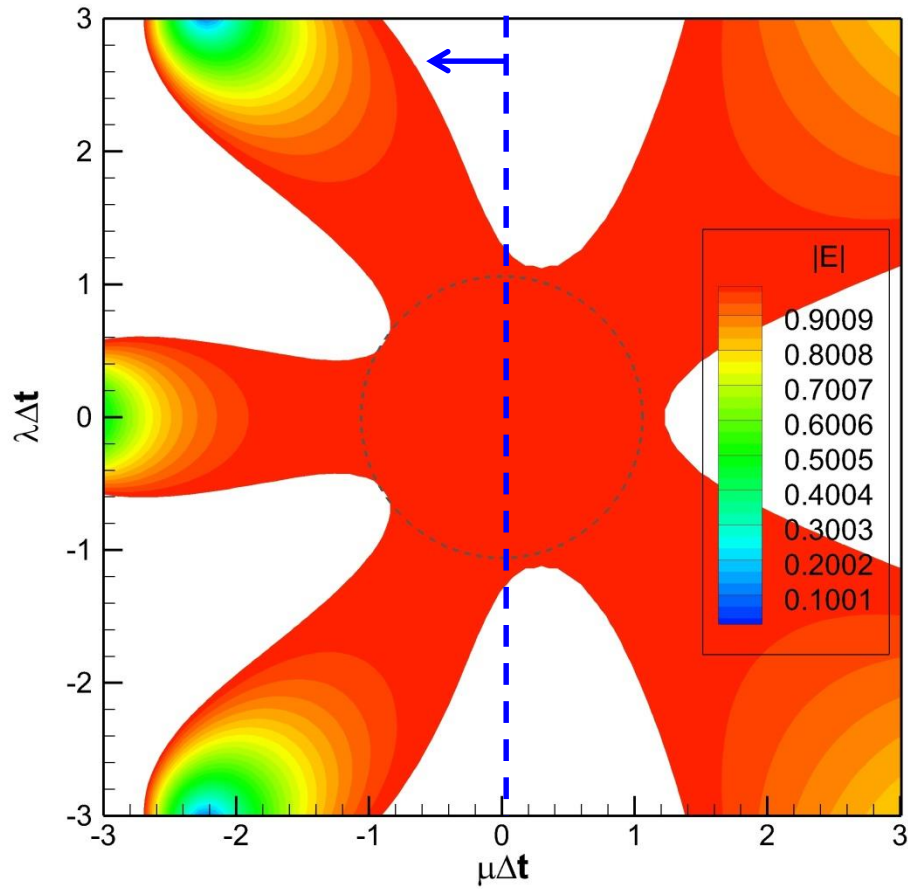
3-stage, 4th order, A-stable

Unstable for CFL > 0.4 for multiple test cases



Dispersion and dissipation errors

Du & Ekaterinaris 2016
4th order, **not** A-stable



Low-dispersion low-dissipation IMRK

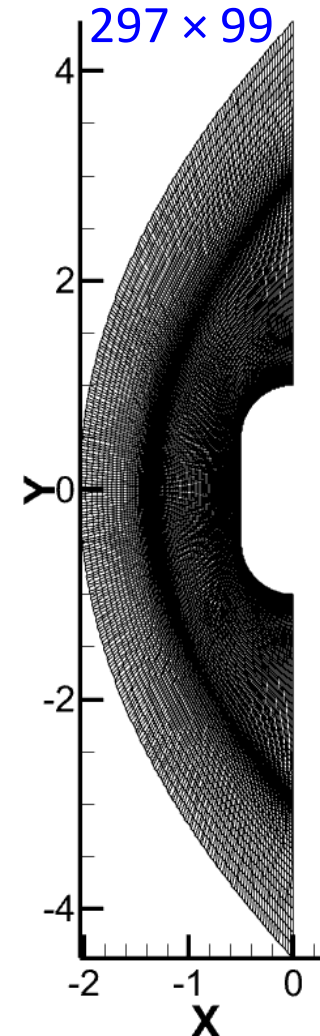
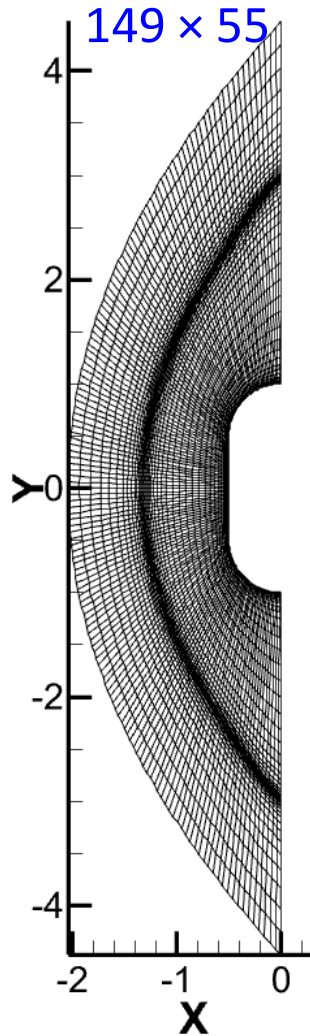
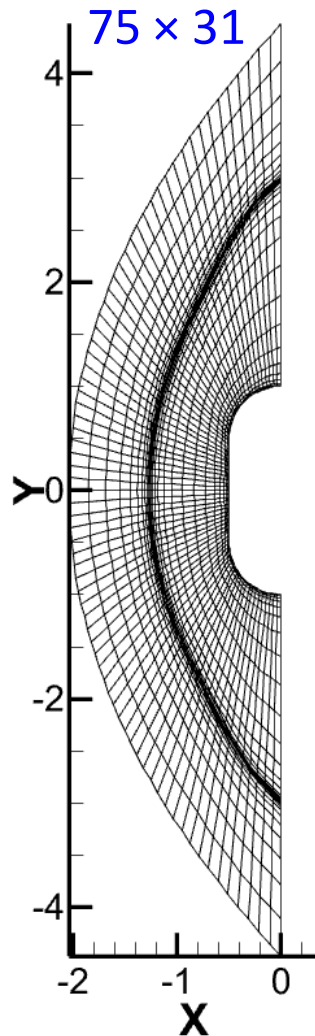
Du & Ekaterinaris (under review), 2016
4th order, **not** A-stable

0.135497354012	0.135497354012		
0.510315342452	0.388552068308	0.121763274144	
0.870923224178	0.271286420495	0.470560027861	0.129076775822
	0.316195414959	0.383754607157	0.300049977884

Inviscid bow shock

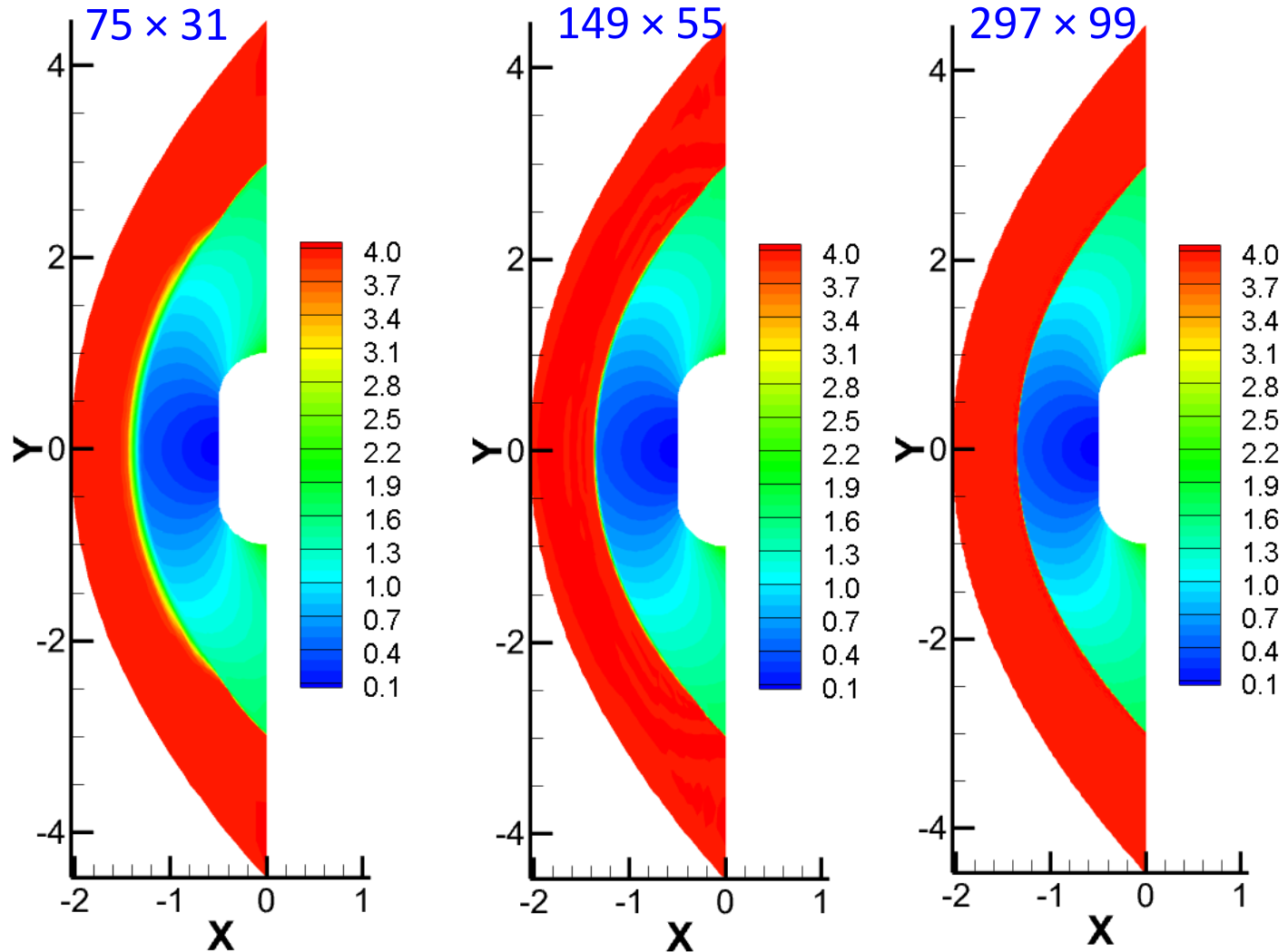
➤ Computational grids

$$Ma_\infty = 4.0$$



Inviscid bow shock

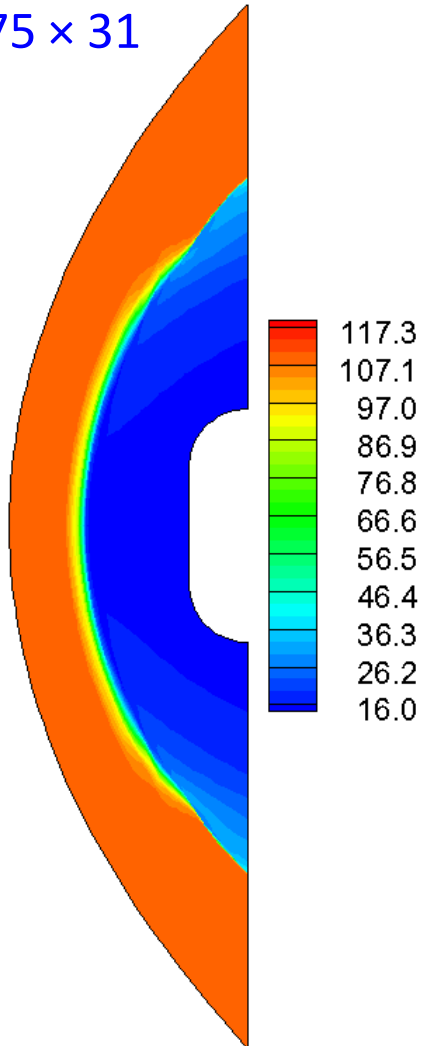
➤ Mach number contours



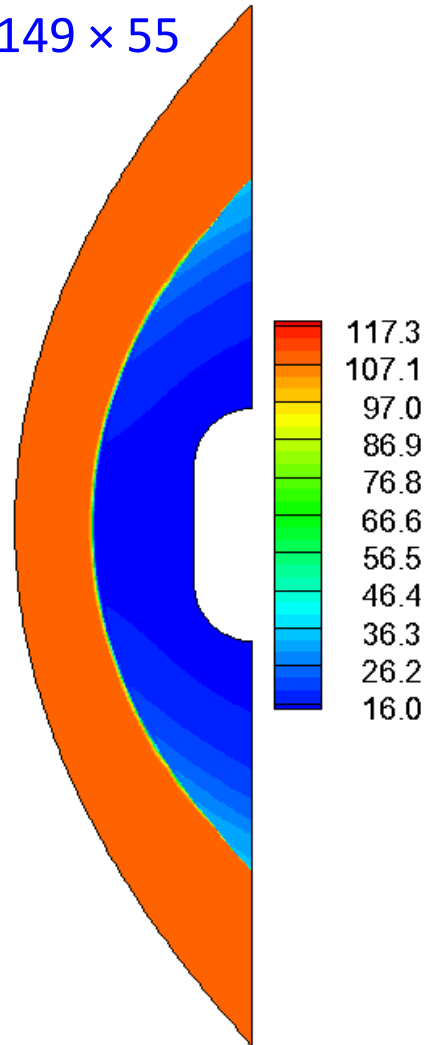
Inviscid bow shock

➤ Total pressure

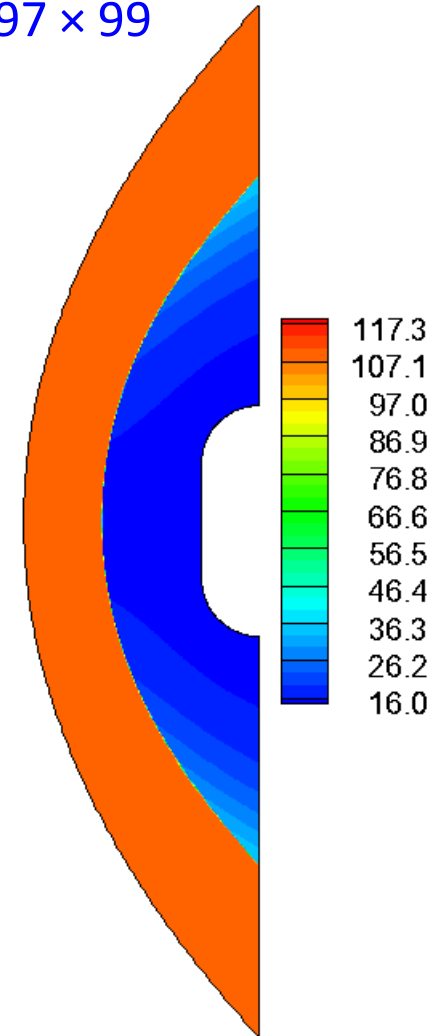
75 × 31



149 × 55



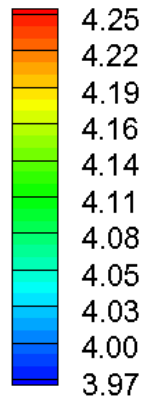
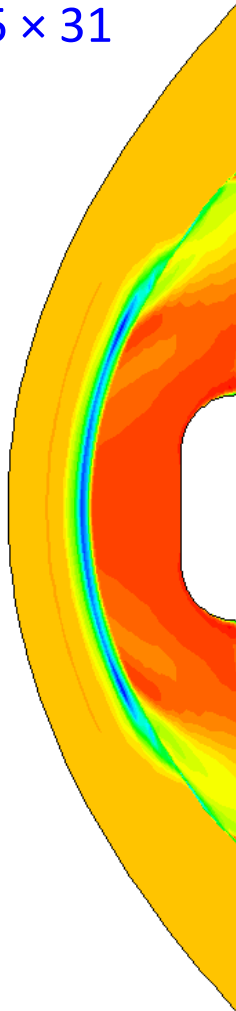
297 × 99



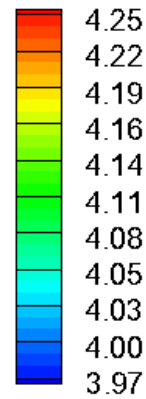
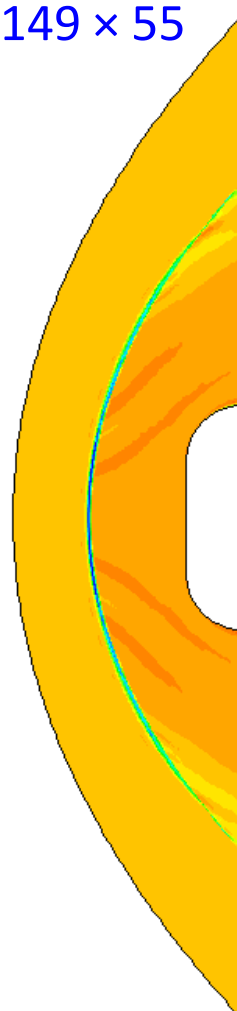
Inviscid bow shock

➤ Total temperature

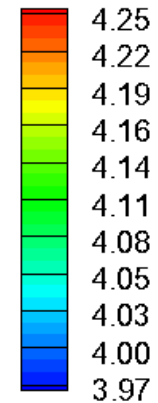
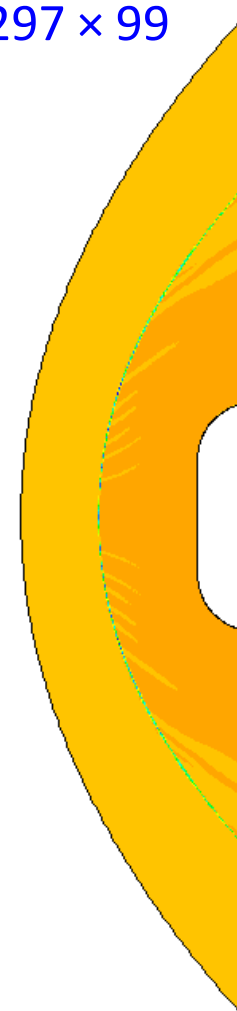
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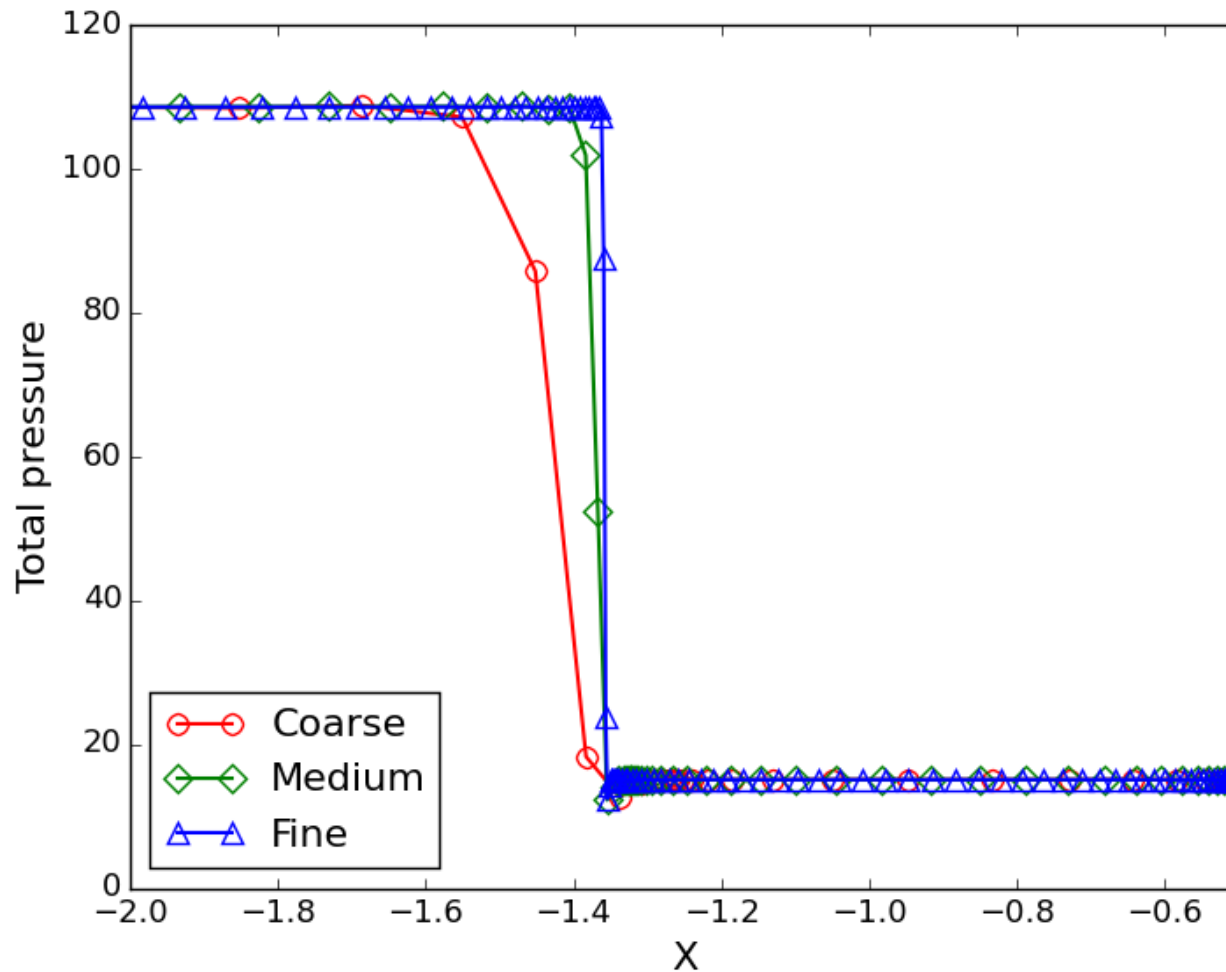


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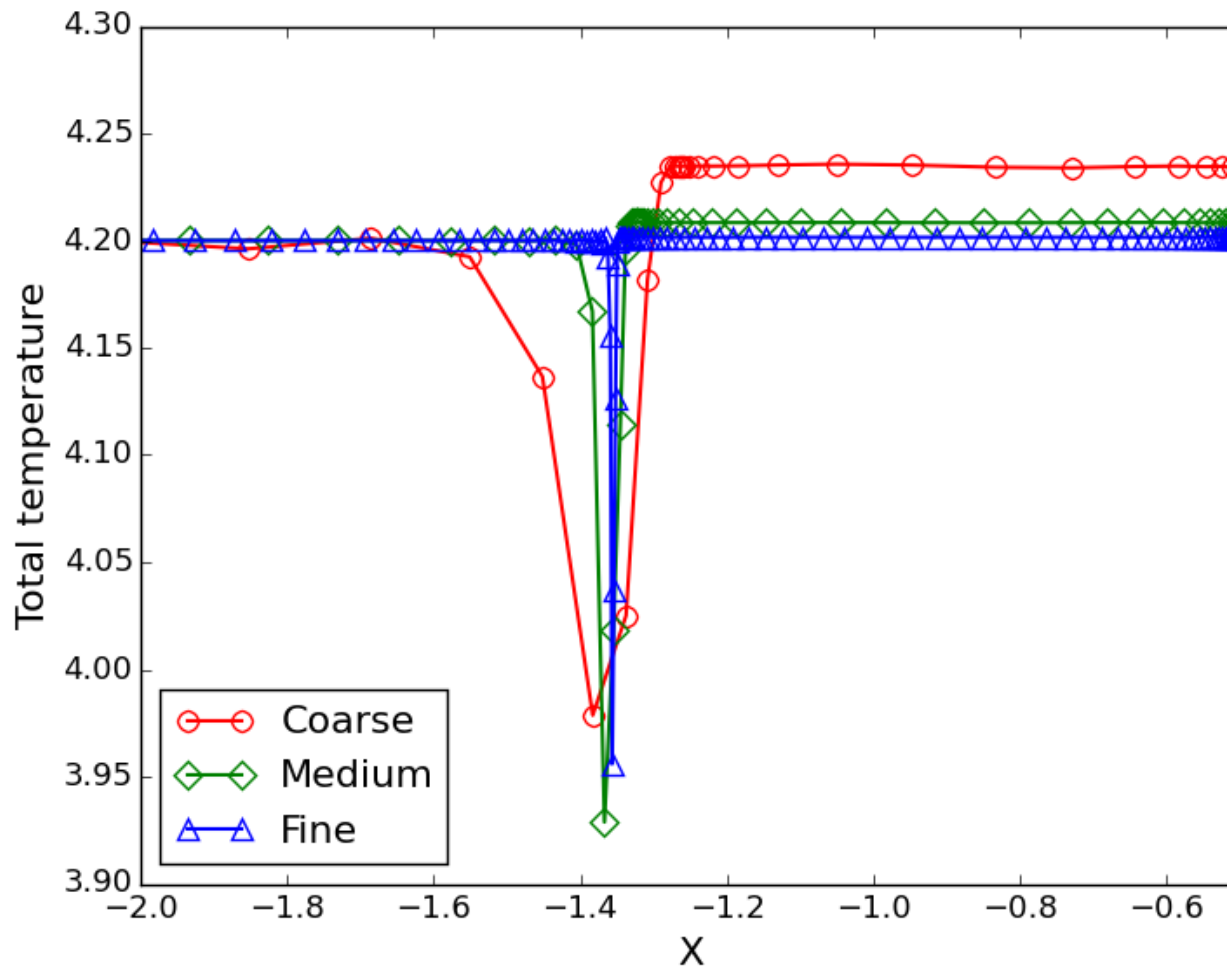
Inviscid bow shock

- Total pressure – symmetry line, $y=0$



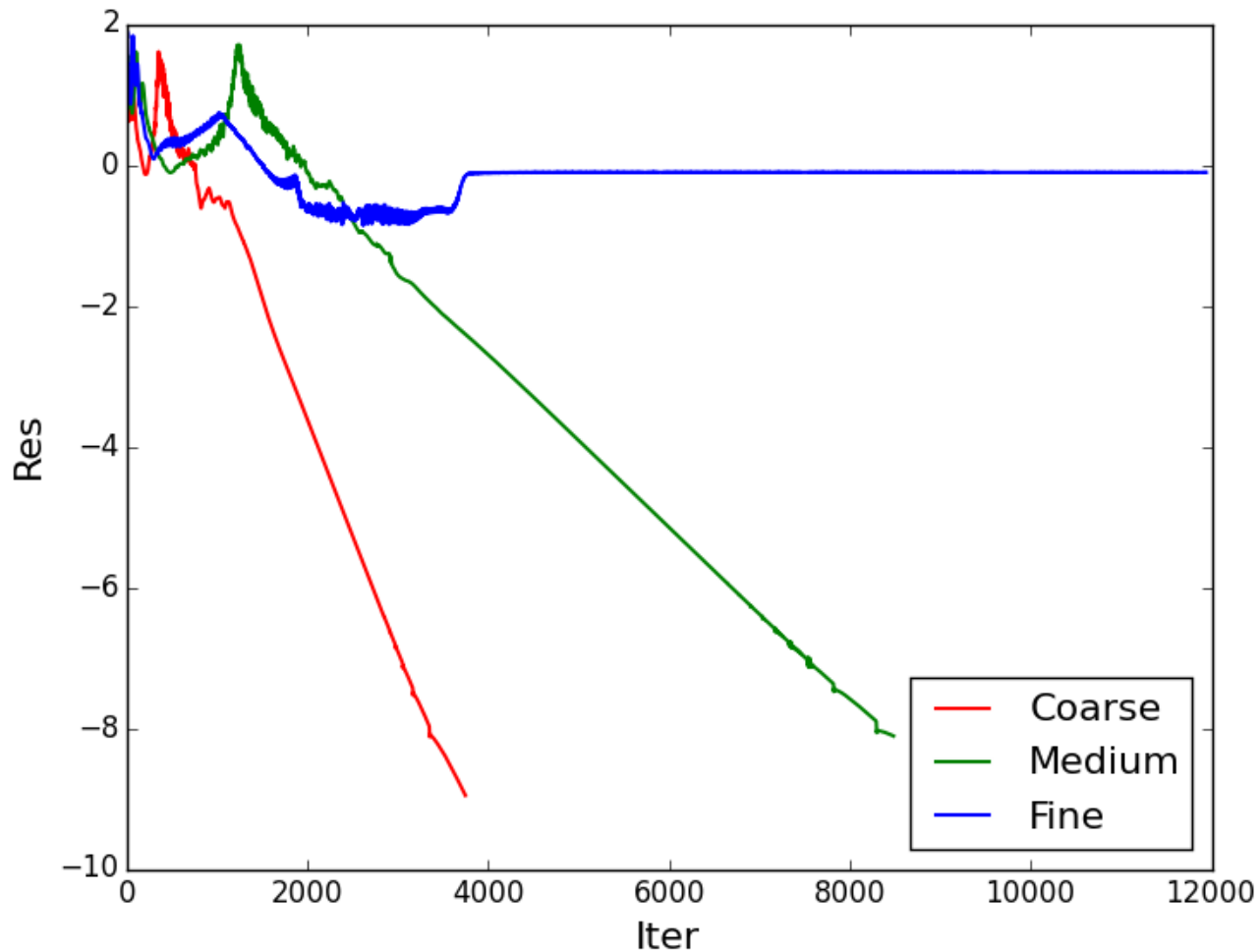
Inviscid bow shock

- Total temperature – symmetry line, $y=0$



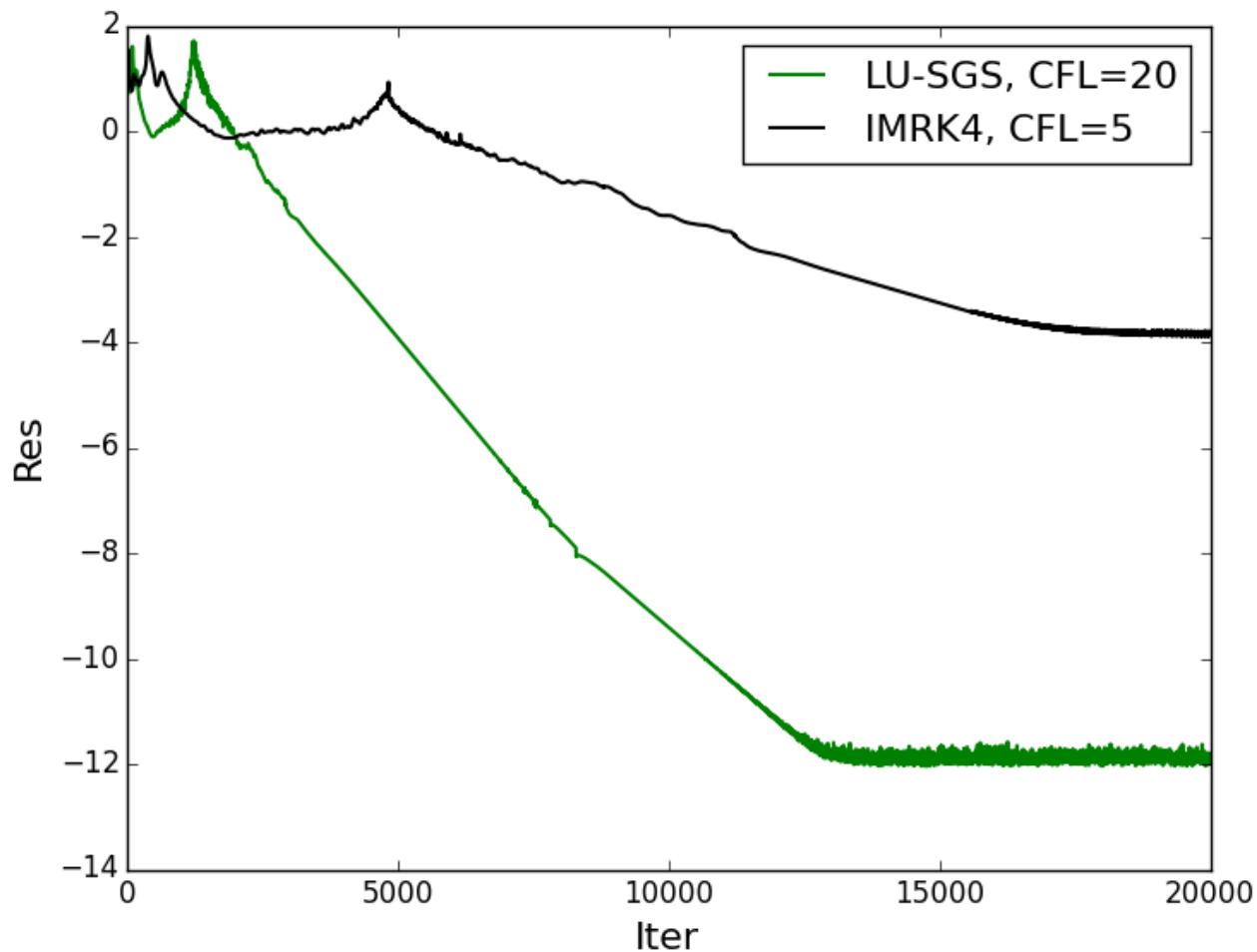
Convergence histories

- LU-SGS, CFL=20, continuity equation



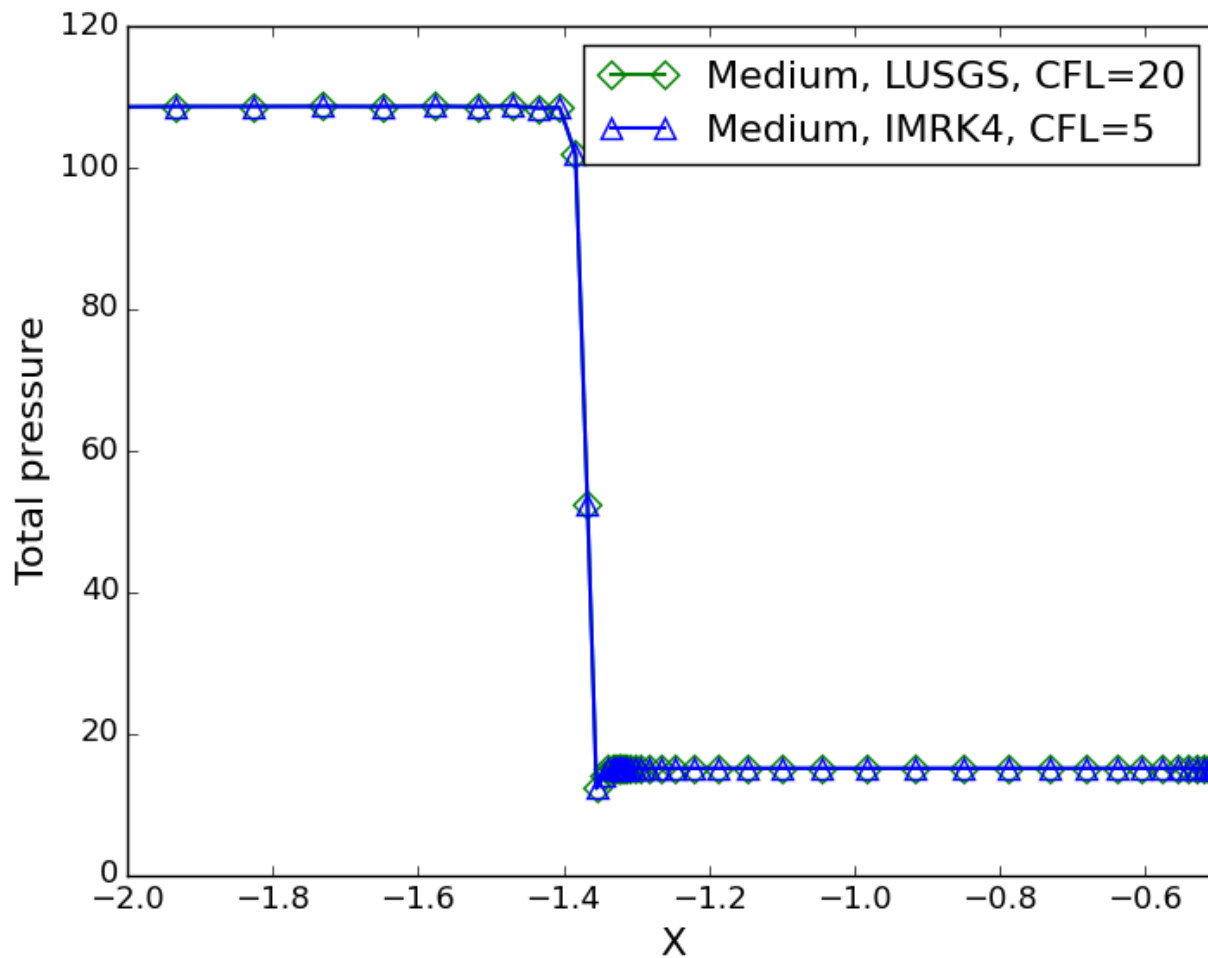
LU-SGS vs IMRK4

- Convergence histories, continuity equation



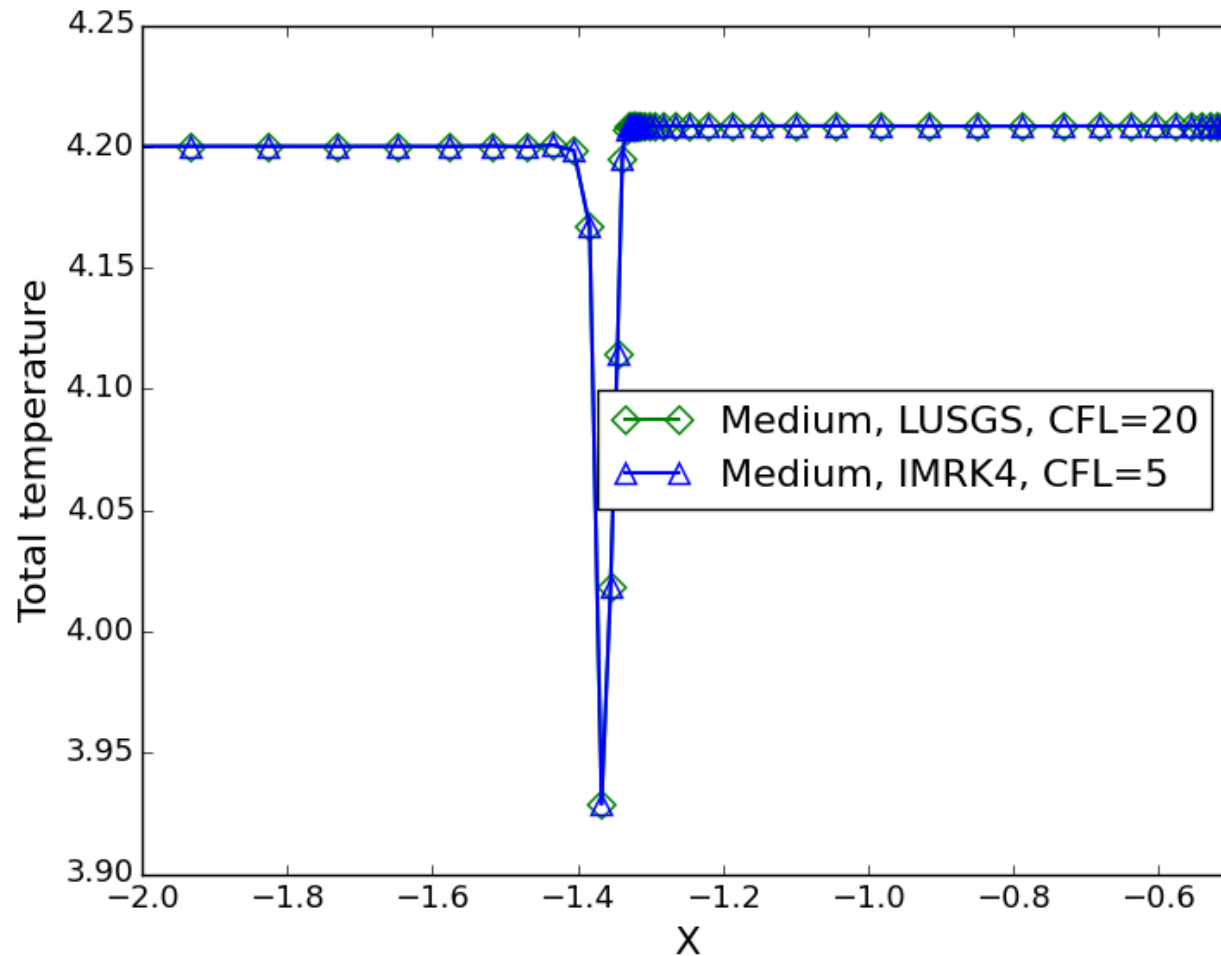
LU-SGS vs IMRK4

➤ Total pressure



LU-SGS vs IMRK4

➤ Total temperature



Summary

- The classical 5th order WENO scheme was applied for the computation of a bow shock at $M = 4$.
- Implicit marching was used for fast convergence to the steady state
- The numerical predictions for all meshes were in fair agreement
- A small amplitude jump of total temperature at the shock was found for all meshes
- A the total temperature before and after the shock remained almost constant for the fine mesh
- A small increase in total temperature was found with coarser meshes